

## Derived Atmospheric Density Corrections from Historical EOS Mission(s) Ephemeris Data for Space Weather Model Validation

September 18, 2014

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## Background



- An unpredicted, post-maneuver close approach between Aqua and a piece of orbital debris required a quick response Risk Mitigation Maneuver (RMM)-10/25/2013
  - RMM required two days after a routine orbit maintenance maneuver was performed
    - Caused an unplanned control box violation (mission CB not science CB)
- This activity resulted in an Earth Science Mission Operations (ESMO) review board which included representatives from the Space Weather group
  - Follow-on discussions with Space Weather representatives resulted in a proposal to investigate the possible use of derived Aqua atmospheric density data for use in space weather model validation



# Agenda



- Aqua Mission Background
- Routine Flight Dynamics Operations
- Definitive Ephemerides
- GSFC Flight Dynamics Facility
- Proposed Approaches for Derivation of Atmospheric Density
- Aqua Magnetometer Data



# EOS Mission Orbit (Aqua)



- Aqua was launched on May 4, 2002
- Ascended into a polar, sun-synchronous, frozen orbit
  - 98.2 degree inclination
  - 705 km semi-major axis (SMA)
  - 16 day repeat cycle
  - Mean Local Time (MLT) range: 1:36:30 +/- 45 seconds
- Maintains World Reference System-2 (WRS-2) ground track to +/-10km (mission goal)
  - WRS-2 was originally established by USGS for cataloging Landsat data
  - 20km is the Aqua ground track maintenance science requirement
  - EOS Flight Dynamics team (FDT) receives a daily Aqua definitive ephemeris from GSFC-Flight Dynamics Facility (FDF)

## Aqua Flight Dynamics Team Routine Operations



- FDT propagates the Aqua predicted orbit from the end of the definitive ephemeris for both short- and long-term duration predicted product generation
  - Generate daily Aqua on-board computer (OBC) ephemeris upload
    - Predicted OBC ephemeris accuracy requirement is 400 meters (RSS) after 32 hours
  - Predicted Aqua OBC ephemeris accuracy requirement (400 m RSS) is occasionally exceeded due to errors in predicted solar flux
    - Predicted ephemeris accuracy depends on the NOAA 27-day solar flux prediction
    - Flux predictions are less reliable during peaks of solar cycle
  - Determine if orbit maintenance maneuvers are necessary to meet
    mission constraints using predicted ephemeris
    - Plan orbit maintenance maneuvers
      - > Drag make-up (DMU) maneuvers solar flux level dependent (density)
      - > Inclination maneuvers compensate for solar/lunar effects on inclination





## Aqua Flight Dynamics Team Routine Operations



- Predicted ephemeris is also used to plan risk mitigation maneuvers (RMMs)
- PROBLEM: Predicted miss distances depend on the accuracy of the predicted solar activity and modeling the effects of density changes on each object
  - Each object has unique physical properties resulting in differential drag effects
  - Improvements in space weather predictions and the effects on atmospheric density would be beneficial for routine maneuver/RMM planning and could also improve predicted ephemeris accuracy
    - RMM planning is based on short-term (<1 week) predictions of the orbits of a primary and a secondary object
    - If the solar flux prediction is not accurate, the predicted miss distance between objects is not accurate either



## Flight Dynamics Facility (FDF) - GSFC



- FDF provides a daily definitive ephemeris for the Aqua (and Aura) missions
  - Nominal tracking schedule for Aqua orbit determination (OD)
    - One ground station pass per orbit (~14/day)
    - 4-6 TDRS passes per day
  - 24 hour tracking arc is used for daily OD
  - 20 meters (RSS) is the required definitive OD accuracy
    - OD accuracy requirement defined as the maximum error over the 24 hour overlap segments in successive FDF definitive products
      - > i.e., today's definitive vs. yesterday's prediction at end of filter run
    - Most of that error is in the along track direction
- The FDF/EOS FDT maintain a historical ephemeris record (since launch) of the definitive orbit solutions and concatenates them into yearly segments





## FDF: Quick Refresher on Drag



$$\vec{a}_{\rm drag} = -\frac{1}{2}\rho \frac{C_D A}{m} v_{\rm rel} \vec{v}_{\rm rel}$$

- $\rho$  = density
  - D = drag coef.

(note: "fitted" vs. "physical")

- A = cross-sectional area
- *m* = mass
- $\vec{v}_{rel}$  = atmosphere-relative velocity

#### Fitted C<sub>D</sub>:

 estimated parameter that is solved for during orbit estimation process  Physical C<sub>D</sub>:
 determined by energy and momentum exchange of atmospheric particles with surface

- depends on geometry, orientation, gas-surface interaction (GSI) model, atmosphere chemical composition, relative velocity, gas temperature, wall temperature
- closed-form solutions exist for simple shapes, but complex shapes need numerical solutions, e.g. Direct Simulation Monte Carlo (DSMC)



# FDF: Relation to Other Density Estimation Efforts?

#### Things like CHAMP, GRACE, GOCE:







Precision measurements (accelerometers, GPS) on small number of satellites
High-fidelity force modeling (required)
High temporal resolution (~sec), local to orbit
Low altitude (<450 km)</li> Things like Aqua (and other EOS):

- Pretty good
  measurements
- Medium-fidelity force modeling

 Medium temporal resolution (~per orbit?), local to orbit

• Upper thermosphere (~700 km) Things like HASDM (High Accuracy Satellite Drag Model) from USAF:

- Radiometric tracking of ~100 targets (e.g. rocket bodies) throughout thermosphere.
- Somewhat well-known drag properties.
- Dynamic Calibration of the Atmosphere (DCA)
  - Spatially-resolved (global correction)
  - Low temporal resolution (~3 hrs).





# FDF: A Proposed Strategy



## Simple approach:

- Estimate only a density correction, where C<sub>D</sub> is calculated from ODTK's box-and-wing model.
- In parallel, estimate both density correction and C<sub>D</sub> correction simultaneously, as done now.
- Cross-validate the results from the above two.
  - Also, validate using predictive vs. definitive ephemeris comparisons.

## **Expanded approach:**

Compare density estimate with other models?

Also:

• Investigate higher-fidelity SRP modeling.



## EOS FD - Proposed Analysis Approach



## Initial Concept:

- Aqua drag area is repeatable on an orbital basis
  - Daily FDF solved-for  $C_D$  variation ranges from ~0.5 to ~3.0 over mission life
- Identify "small" predicted vs. definitive compares (e.g. <50m over the predicted interval)
- Look for consistency in the C<sub>D</sub> value(s) used in the predicted ephemeris when the compares were <50m</li>
  - If the predicted vs. definitive compare was small, that implies that the C<sub>D</sub> value used to generate the predicted ephemeris was correct for the atmospheric model used
  - If the C<sub>D</sub> values were consistent for the small compares, then use this value as the true C<sub>D</sub> value for correcting the predicted atmospheric density model used by the FDT
- FDF can output definitive OD solutions at orbital frequency to 20 meter accuracy
  - FDF can also output the solved-for  $\mathbf{C}_{\mathrm{D}}$  , predicted density, and solved-for density







# EOS Mission Magnetic Field Information



- Aqua (and Aura) use three-axis magnetometers (TAM) for coarse attitude determination/control and for momentum management
  - Post-calibration magnetometer accuracy should be within a few milli-gauss per axis
  - Magnetic field strength and direction at 705 km can be derived from this information
    - Total magnetic field strength is ~ 300 milliGauss at 705 km
    - 16-day repeat cycle over WRS-2 grid
    - <20 meter spacecraft position accuracy available when using definitive ephemeris</li>
    - TAM output available at 32 second intervals
  - Historical magnetometer record is available (with effort by the Flight Operations Team [FOT]) from near launch for both Aqua and Aura
    - Sample Aqua TAM data were provided by John Nidhiry (Aqua FOT GN&C Lead)



## EOS Magnetic Field Information



- TAM data are potentially useful for Space Weather model validation/enhancement
  - Historical data are available from other EOS missions, if Aqua magnetometer data are shown to be useful

### Aura Location

- Aura crosses the ascending node ~8 minutes behind Aqua
  - Same WRS-2 path as Aqua, but offset ~18 km East of Aqua
    - > To facilitate coincident observations between CALIPSO CALIOP and Aura MLS instruments

