

#### NAIRAS Version 3.0

### Atmospheric/Geospace Ionizing Radiation Environment Model

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### Projects/Team Members

- 1. Improvements to Solar Energetic Particle (SEP) Dose Predictions (NASA Space Weather Science Applications Program)
  - NASA Langley Research Center (LaRC): Chris Mertens, Guillaume Gronoff, Daniel Phoenix
  - West Virginia University: Piyush Mehta and Smriti Nandan Paul
  - NASA Goddard Space Flight Center's Community Coordinated Modeling Center (CCMC): Yihua Zheng
- 2. Commercial Crew Program (CCP) Post-Flight Reference Radiation Environments (NESC)
  - NASA LaRC: Chris Mertens, Guillaume Gronoff, Daniel Phoenix
  - NASA Marshall Space Flight Center: Joe Minow and Emily Willis
  - NASA Kennedy Space Center: Janessa Buhler
  - Jet Propulsion Laboratory: Insoo Jun
  - CCMC: Yihua Zheng



#### Nowcast of Aerospace Ionizing RAdiation System (NAIRAS) Model

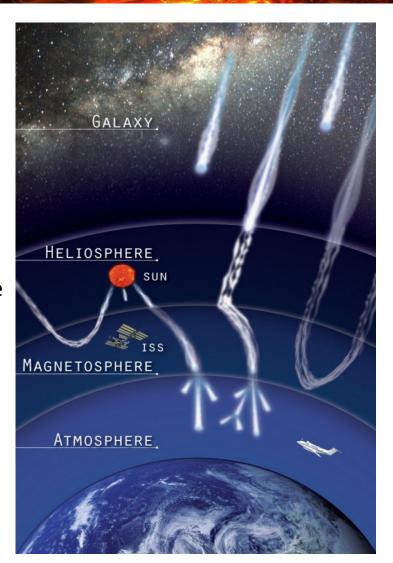
- Running in real-time on LaRC computer cluster since 2011, results hosted on Space Environment Technologies server/website
- Running in real-time at CCMC since 2020

#### Key Model Features

- Global atmosphere ionizing radiation environment model
- Physics-based **HZETRN** (High Charge (Z) and Energy TRaNsport) code
- Real-time inclusion of solar energetic particle (SEP) radiation
- Real-time solar-magnetospheric effects on radiation (cutoff model by Kress et al. [2004, 2010])

#### New/Current Model Development

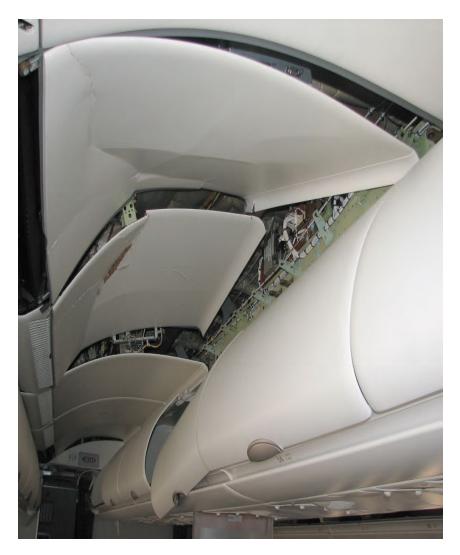
- Improved SEP dose nowcast and forecast
- Extend to low-Earth orbit (LEO) environment
- Single-Event Effects (SEE) radiation risk assessment quantities
- Run-on-Request (RoR) @ CCMC





### **Aviation Radiation Avionic Effects**

- Cosmic radiation effects on Avionics Systems
  - Interaction with semiconductor material, depositing charge causing single event effects (SEE) → change in logic state
  - Number of recorded instances of avionic SEE at GCR exposure levels (e.g., Normand et al., 1997, 2001; Olsen et al., 1993)
  - SEE in autopilot systems correlated with CR flux (altitude and latitude variation)
  - Avionics SEE occurrence rate (Royal Academy of Engineering, 2013)
    - o GCR: every 200 flight hours
    - Solar storm: > 1 per hour (scaled Feb 1956 event)
  - Near catastrophic event: Qantas Airways flight 72,
     October 7, 2008 (pictured right)
    - SEE most probable explanation. All other environmental causes ruled out (ATSB, 2011)
    - Intermittent, incorrect inertial reference data initiated violent pitch-down command from flight control system
    - 110/303 passengers and 9/12 crew injured; 12 occupants seriously injured; 39 received hospital treatment
- For aircraft systems (as opposed to components) radiation standards and industry awareness less developed
  - Guidance standards only
  - No regulatory standards 06/09/2022



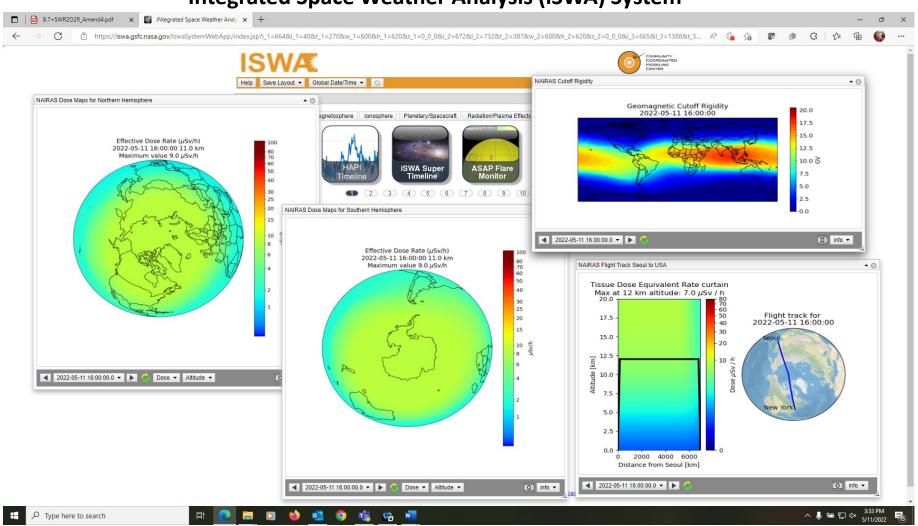


- NAIRAS Real-Time Interface @ CCMC (publicly available)
- NAIRAS RoR Capability @ CCMC (coming soon)
  - Model updates and improvements
  - Expanded output products
  - LEO orbit example
  - Comparison to NASA Radiation Dosimetry Experiment (RaD-X) balloon flight measurements
- SEP Improved Nowcast and Forecast Developments (coming soon and under development)
  - Geomagnetic cutoff rigidity
  - SEP Proton Spectral Fitting
- Summary and Conclusions



### Real-Time NAIRAS @ CCMC

#### **Integrated Space Weather Analysis (iSWA) System**





### NAIRAS Model Improvements

- LEO radiation environment (trapped protons)
- Extend GCR model to ultra-heavy nuclei (Z=29-92,A=64-238) for SEE assessment from high linear energy transfer (LET) processes
- RoR Capability
  - Output: (1) global dosimetric quantities and (2) flight trajectory dosimetric and flux/fluence quantities
  - Differential/integral flux/fluence quantities useful for SEE assessment
  - Generic input flight trajectory capability (aircraft, balloon, spacecraft)
  - Improved atmospheric transport: off-zenith directions included
- Expanded geomagnetic cutoff rigidity model to use either TS05 (previous version) or T89 magnetospheric magnetic field models
- Improved SEP proton spectral fitting to address
  - Representing relativistic protons during ground level enhancements (GLEs)
  - Overall algorithm robustness in real-time operation



### NAIRAS Results for LEO Trajectory

 NAIRAS Total Trajectory Effective Dose (per day)

○ **GCR**: 215 uSv

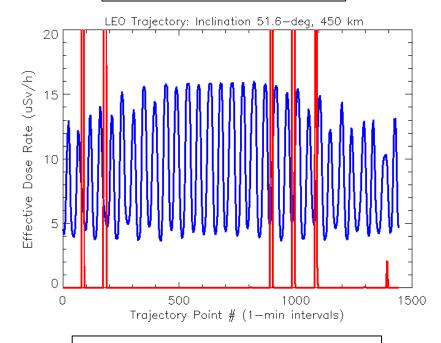
Trapped proton (TRP): 163

uSv

○ Total: 378 uSv

- International Space Station (ISS) Total Effective Dose (per day)
  - GCR: 233 uSv (Wu et al., 1996)
  - TRP: 166 uSv (Wu et al., 1996)
  - Total: 438 uSv (Cucinotta, 2008)

Nov 02, 2003 16:00 UT to Nov 03, 2003 16:00 UT



**Blue:** GCR effective dose rate

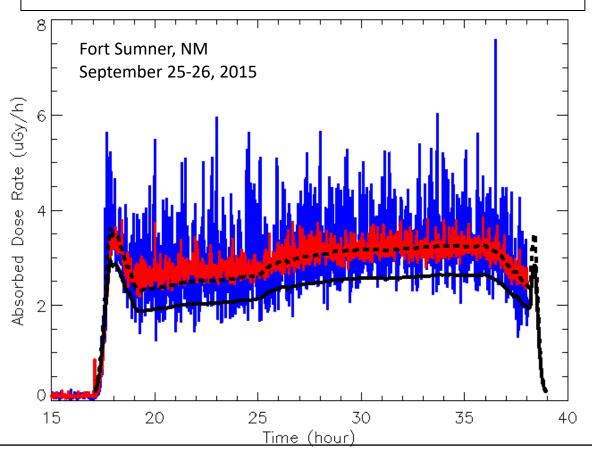
**Red**: TRP effective dose rate

Al Shielding: 50 g/cm<sup>2</sup>



### NASA RaD-X Balloon Flight

#### Time Series of Dose Rates Measured on RaD-X Balloon



Liulin; TEPC; NAIRAS Ti-Dose (Dashed); NAIRAS Si-Dose (Solid)

**Region A** (Balt: 21-27 km) **Diff** = -0.2% | **Region B** (Balt: > 32.5 km) **Diff** = -8.4%

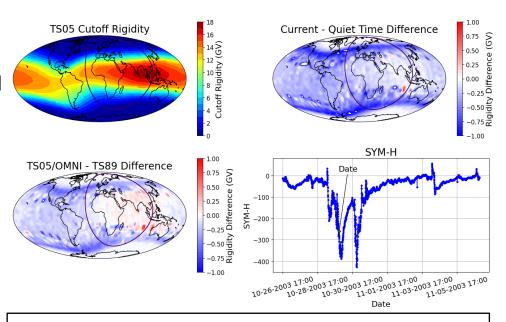


### Geomagnetic Cutoff Rigidity Model

- Based on CISM-Dartmouth model with TS05 magnetospheric B-field (Kress et al., 2010)
- Added multiple magnetospheric Bfield selection capability
  - TS05 → parameterized by solar wind quantities, interplanetary magnetic field (IMF), SYM-H/Dst, and other derivative solar wind quantities
  - T89 → parameterized by the planetary K-index (Kp)
- The TS05 better represents magnetospheric responses to interplanetary disturbances
  - but real-time solar wind parameters available from ACE/DSCOVR 1995+
- Benefits of T89 option
  - NAIRAS can simulate any historical solar-geomagnetic storm event
  - Extend/enhance validation capabilities
  - Provide initial step in forecasting cutoff via Kp-parameter forecast 06/09/2022

#### Halloween 2003 Geomagnetic Storm

Date: 10/29/2003 2100 UT



**Top Right**: Largest suppression of cutoff (~1 GV) (openclosed field boundary) occurs in dusk sector due to max build-up of partial ring current in TS05 (IMF Bz dependent)

**Bottom Left**: T89 doesn't well represent max cutoff suppression and cutoff in dusk sector

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### Machine Learning Kp/Dst-Forecast

#### Kp/Dst-Forecast Approach

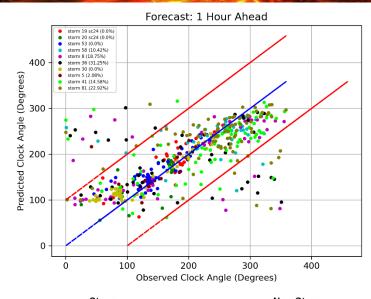
- WSA-ENLIL-Cone solar wind parameters forecast
- Empirical formula to get Kp/Dst as function of solar wind speed, total IMF and IMF clock angle (Newell et al., 2007)
- However, need separate IMF clock angle forecast to improve state-of-art (@CCMC) since WSA-ENLIL-Cone has no internal coronal mass ejection (CME) structure

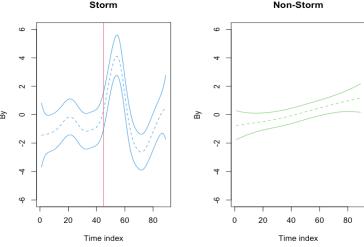
#### Machine Learning IMF Clock Angle

- Trained on Advanced Composition Explorer (ACE) data (solar wind velocity and density, IMF Bcomponents, derived clock angle) from large geomagnetic storms (Dst min < -100 nT) during solar cycles 23 and 24
- Developed deterministic and stochastic models
- Forecast 1-12 hours ahead

#### Key Results

- IMF clock angle predictions provide improvement over current operational Kp/Dst models at CCMC (top right). However, beyond the first couple hours the performance is unacceptable
- Improved performance sought using Functional Data Analysis (FDA) methods (bottom right)





Functional means with 95% uncertainty bands. Vertical line marks storm onset

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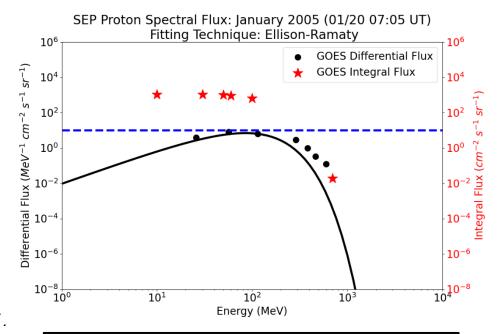
### Update to SEP Spectral Fitting

#### New Approach

- Fit spectrum to Geostationary Operational Environmental Satellites (GOES) integral proton flux rather than differential flux measurements
- Fit four functional forms to GOES integral proton
   flux
- Choose solution with minimum chi-square

#### Benefits

- Improved robustness
  - Difficulty fitting GOES differential channels at event onset and for weak-to-moderate events
  - Extrapolation beyond highest differential energy channel (~500 MeV) requires introducing arbitrar and subjective criteria
  - 50% or more of SEP effective dose at large material depths (aviation altitudes) comes from > 500 MeV protons
- Preliminary simulations using neutron monitor data suggest fitting to GOES integral proton flux may better represent the relativistic protons during GLEs
- New integral flux fitting approach provides a pathway to develop a SEP proton spectrum forecast



SEP proton spectrum (black line) fit to GOES integral flux and comparison to GOES differential proton flux. Horizontal blue line indicates NOAA/SWPC SEP event threshold for >10 MeV proton flux.



### New SEP Spectral Fitting Algorithm

#### **SEP proton spectral fitting problematics**

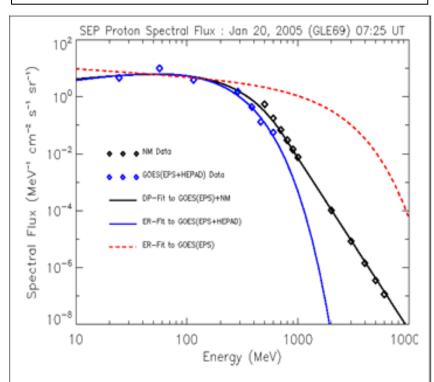


Figure 1: GOES (EPS +HEPAD) differential proton flux measurements and NM-inferred differential proton flux for January 20, 2005 SEP/GLE. Double power-law (DP-Fit) and Ellison-Ramaty (ER-Fit) functional fits to the observations.

#### New approach using GOES integral flux

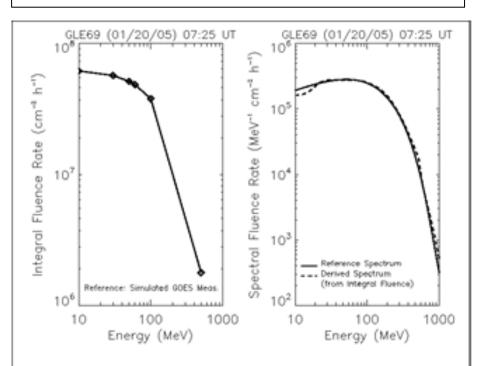


Figure 1: (left) Simulated GOES integral flux measurements (diamonds). (right) Results of new spectral flitting algorithm (dashed) compared to reference spectrum (solid) in previous figure.

- Major NAIRAS Code Deliverables to CCMC/iSWA
  - NAIRAS Real-Time Global Dosimetric Quantities (Publicly Available Now)
  - NAIRAS RoR Capability (Publicly Accessible in August 2022)
  - NAIRAS Improved SEP Proton Spectral Fitting Algorithm (Operational in Fall 2022)
- Significant Improvements to NAIRAS Model: Developed, Implemented and Tested
- NAIRAS predicts both dosimetric quantities to assess human radiation exposure and differential/flux quantities to assess SEEs in avionic system
- SEP Dose Forecast Development
  - Geomagnetic Cutoff Rigidity Forecast Model (Under Development)
  - SEP Proton Spectrum Forecast (Begin soon!)



# Backup Slides



#### 1. Global Atmospheric Dosimetric Quantities

- Dose rate products: absorbed dose in silicon, absorbed dose in tissue, dose equivalent, ambient dose equivalent, and effective dose
- Model grid: 1 x 1 lat/lon, 0-90 km @ 1km increments, and 1-hour time cadence
- Input: Start/End Date-Time
- Application: global context and situational awareness of the atmospheric radiation environment; enable retrospective analysis and verification and validation of the real-time version of the NAIRAS model



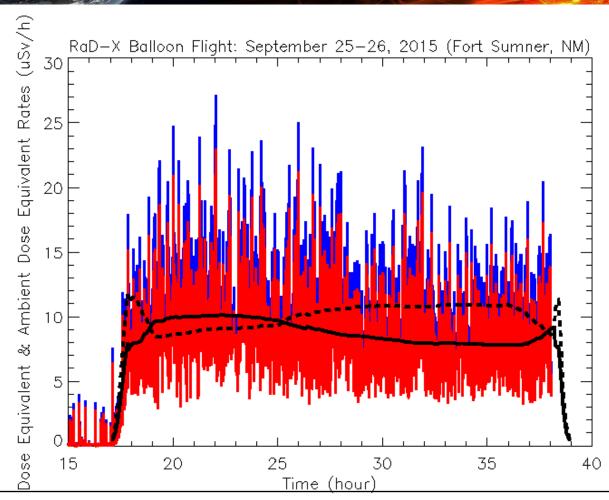
### NAIRAS RoR Output Products

## 2. Trajectory Dosimetric, Differential and Integral Flux and Fluence Quantities

- Dose Quantities (same as for global products)
  - Dosimetric quantities at each trajectory point
  - Time-integrated dosimetric quantities
- Integral Flux and Fluence Quantities
  - GCR LET and trapped/SEP proton flux/fluence
  - Input: lower LET/energy bounds of integral quantities
- Differential Flux and Fluence Quantities
  - GCR LET and trapped/SEP proton flux/fluence
- Input: trajectory file, separate set of shielding depths for dosimetric and flux/fluence quantities
- Application: detailed flight analysis and radiation environment characterization of individual microelectronic components and SEE assessment



### NASA RaD-X Balloon Flight

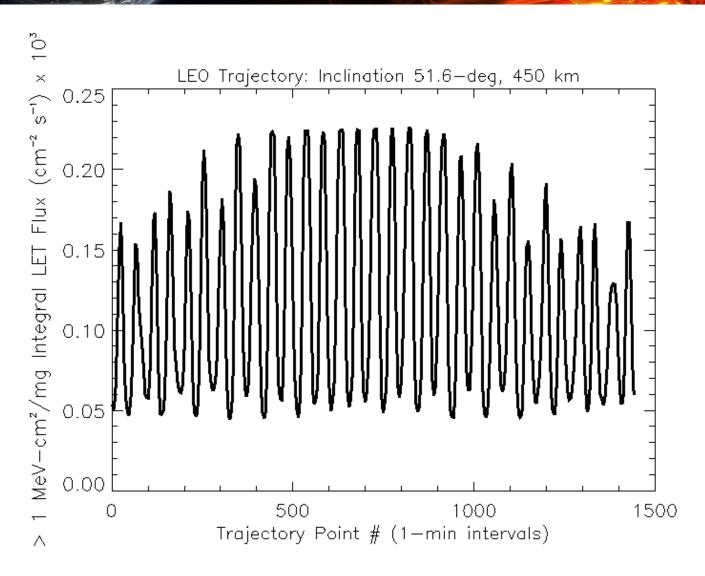


TEPC H\*(10); TEPC DoseEq; NAIRAS H\*(10) (Dashed); NAIRAS DoseEq (Solid)

**Region A** (Balt: 21-27 km) **DEq Diff** = 3.9% | **Region B** (Balt: > 32.5 km) **DEq Diff** = 5.2%

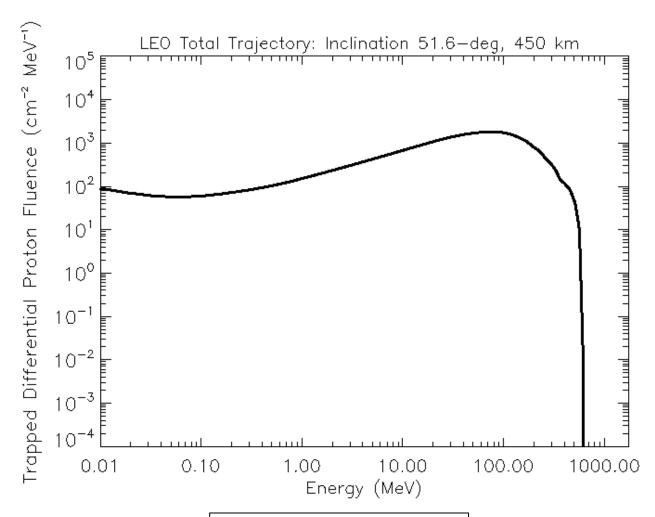


### Trajectory Integral LET Flux





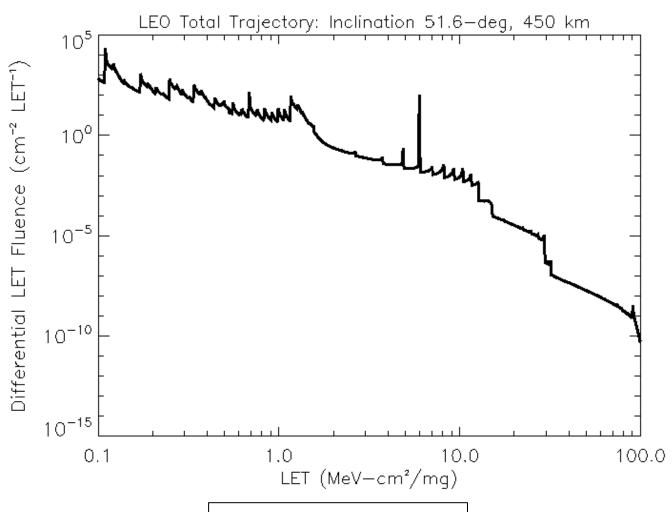
### Trajectory Trapped Proton Fluence



Shielding: 4 g/cm<sup>2</sup> Al-Eq



### Trajectory Differential LET Fluence

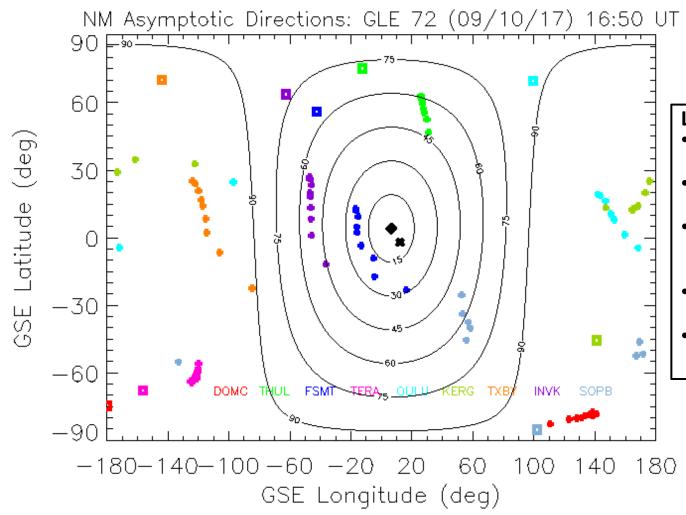


Shielding: 4 g/cm<sup>2</sup> Al-Eq



### Neutron Monitor (NM) Analysis

#### **Asymptotic Directions and Pitch-Angle Distribution**



#### Legend:

- Color Squares: NM Locations
- Color Plus: NM
   Asym Dir (1-5 GV)
- Black Diamond: SEP Proton Asymmetry Direction
- Black Asterisk: IMF Direction
- <u>Contour</u>: Proton Pitch-Angle (deg)